

3- 가

**Analysis of the loss of coolant accidents for the PWR Fuel Test Mode of the 3-Pin Fuel Test Loop**

150

가 3- 가 .  
 MARS 1 1  
 , 1264K 가 가

**ABSTRACT**

Thermal hydraulic analyses have been carried out for the PWR fuel test mode, in order to predict the performance of the emergency cooling water system of the 3-pin fuel test loop for the postulated loss of coolant accidents. The best-estimate thermal hydraulic analysis code of MARS has been used for the analyses. The slit type and double-ended guillotine breaks have been considered at the hot and cold legs in the HANARO pool and the room 1. The present results indicate that the maximum peak cladding temperature is 1264K for the small break LOCA at the cold leg in the room 1. The maximum peak cladding temperature is sufficiently lower than the criterion of the PWR fuel design. Consequently, it is found that the emergency cooling water system is appropriate to remove the heat of the PWR test fuel for the loss of coolant accidents of the 3-pin fuel test loop.

1.

30MW ,

[1]. 1995

. 3-



가

2

1  
가

2.3

3-

, , , 가

가

가

. 가

가

3-

가

가

3.

3.1

3- MARS  
 (Multi-dimensional Analysis of Reactor Safety) MARS  
 RELAP5/Mod3 1 3 COBRA-TF  
 [2,3]. 3  
 MARS 1

3.2

3 3-  
 MARS  
 가 가  
 MARS

7 Sub-volume Pipe  
 heat structure

IR1  
 1 1 1.5  
 $1.2 \times 10^{-5} \text{m}^2$   $9.5 \times 10^{-5} \text{m}^2$   
 2.5 0.2  
 5 가  
 $1.25 \text{m}^3$   $50^\circ\text{C}$   $3.85 \text{m}^3$   
 가 1  $123.23 \text{m}^3$  1  
 가  
 $81.7 \text{m}^3$   
 $40^\circ\text{C}$  가

3.3

2 3 가 700mm  
 9.5mm 13.8mm 가 [4].  
 MCNP 가  
 MARS  
 7 11

30.0kW/m , 41.6kW/m . 1.387 .

0.93 0.7 가 .

ANS 1.2 ,

가 .

(fuel transport and support legs), heat structure . (flow divider) . MCNP .

3.4

MARS .

2

2

5%

5%

2%

2%

3

3

가

가

가

0.615

0.41

0.05

4.

가

가

가 .  
가 .

가 .  
(double-ended guillotine break)

가  
가 (T)

가  
0.01 가  
0.1, 0.33, 0.67, 1.0 (discharge coefficient)  
20% 1%

가  
4.1 1  
1

4

3 4.1 가 가 5.1  
918K 5 1

1% 6%  $5.454 \times 10^{-5} \text{m}^2$  가  
1264K 가 2200°F(1477.6K)  
231 THTF Blowdown RELAP5/Mod.3 (94K) [5].  
1

0.02  
0.094

가  
가  
1  
0.636 1.636

0.436  
0.636 2.936 가

6, 7, 8 1

가

6

가 ,

. 가

가

가

7 1

가

가

5

가

가

가

5

8 1

1

가

3

volume 150( )

volume 445( )

가

가

95

가

가

4.2

가

가

가

9

가

60

100

6%  $2.784 \times 10^{-5} \text{m}^2$

1173K

1

91K

2.1

0.034

0.206

가

가

가

0.65

1.65

1

0.444

0.644

2.944

가

10, 11, 12

가

10

1

, 가

가

가

11

0.5

2

가 0.1kg/s

가

가

가가

11

80

110

0.05kg/s

40

가 0.35kg/s

12

가

가

가

0.9

가

가

5.

가

MARS

1

가

가 . 가  
30.0kW/m 41.6kW/m  
1  
가 ,  
918K .  
가 .  
1  
가 . 1  
1264K ,  
1173K .  
1264K , 가 2200°F(1477.6K)  
3-  
가 가 .

1. “ ,” KAERI/TR-710/96, Korea Atomic Energy Research Institute, Daejeon, 1996.
2. W. J. LEE et. al., “Development of a Multi-dimensional Realistic Thermal Hydraulic System Analysis Code, MARS 1.3 and Its Verification,” KAERI/TR-1108/98, Korea Atomic Energy Research Institute, Daejeon, 1998.
3. W.J. Lee, et. al., “Improved Features of MARS 1.4 and Verification,” KAERI/TR-1386/99, Korea Atomic Energy Research Institute, 1999.
4. C. G. Seo, “Linear Heat Generation Rate of PWR 3-Pin at In-Pile Test Section in IR1 Hole,” Memorandum, HAN-RR-CR-031-03-016, 2003.
5. B. D. Chung, Y. J. Lee, T. S. Hwang, W. J. Lee, and S. Y. Lee, “Quantification of Reactor Safety Margins for Large Break LOCA with Application of Realistic Evaluation Methodology,” J. of KNS, Vol. 26, No. 3, pp. 355-366, 1994.

1. 3-

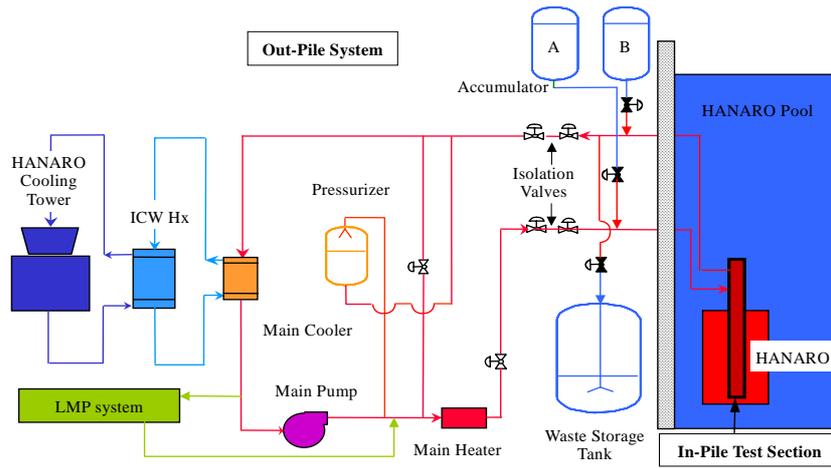
		가	
	(MPa)	17.5	17.5
	(°C)	350.0	350.0
	(kW)	112.3	116.2
	(°C)	300.3	276.7
	(MPa)	15.6	10.1
	(kg/s)	1.6	1.63
	(m/s)	4.6	7.2

2.

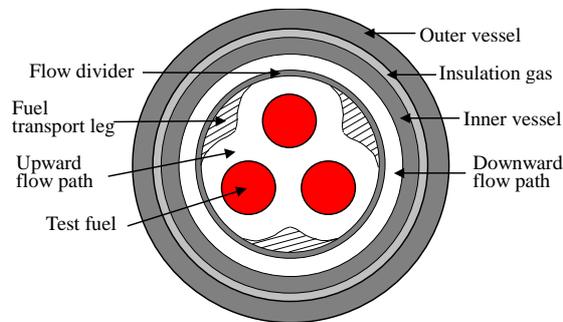
	MARS	(%)
(kW)	66.15	105
(kW)	47.04	105
(kg/s)	1.52	95
(MPa)	15.288	98
(K)	306.3	102
(MPa)	15.794	95

3.

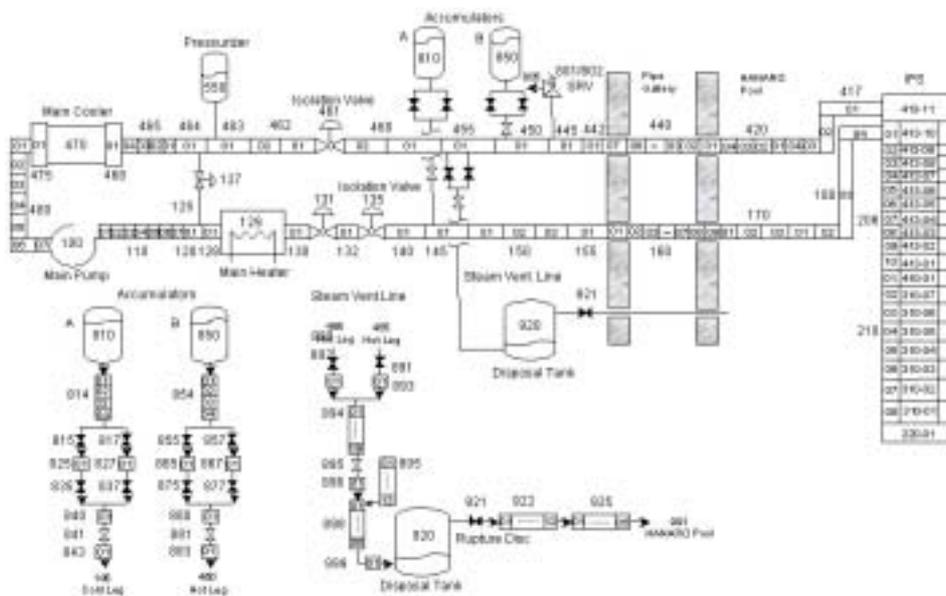
	(kg/s)	1.28
	(kg/s)	1.84
	(°C)	325.0
	(MPa)	14.13
	(MPa)	17.24
	(MPa)	9.0
-	(kg/s)	0.96
-	(kg/s)	1.84
-	(°C)	325.0
- /	(MPa)	13.44



1. 3-

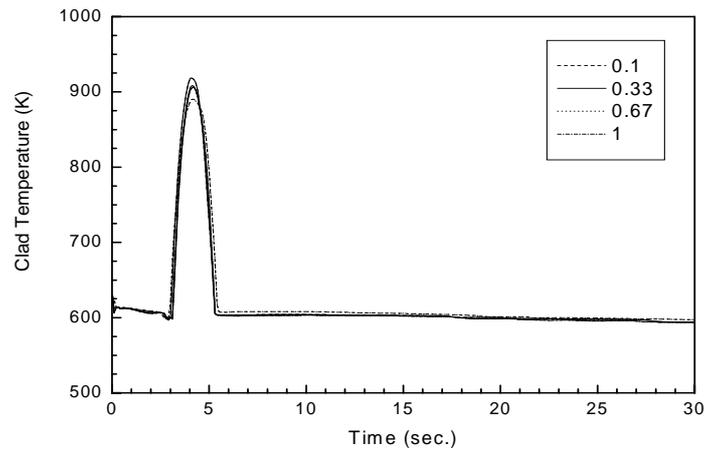


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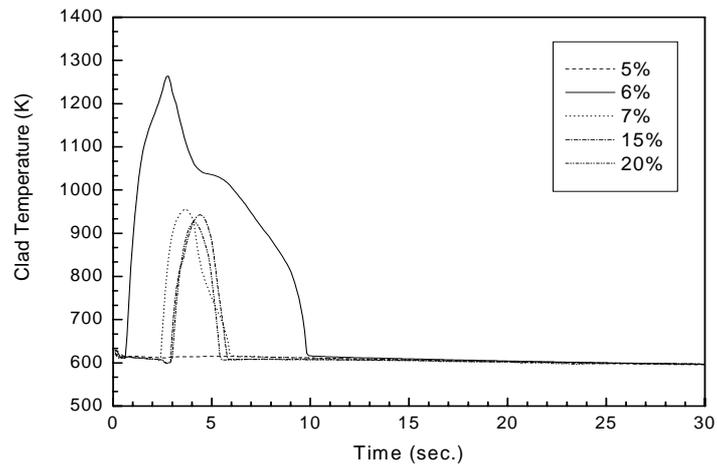


3. 3-

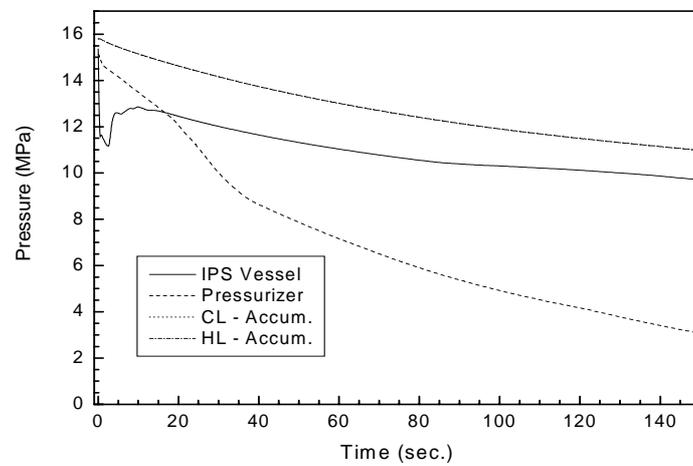
MARS



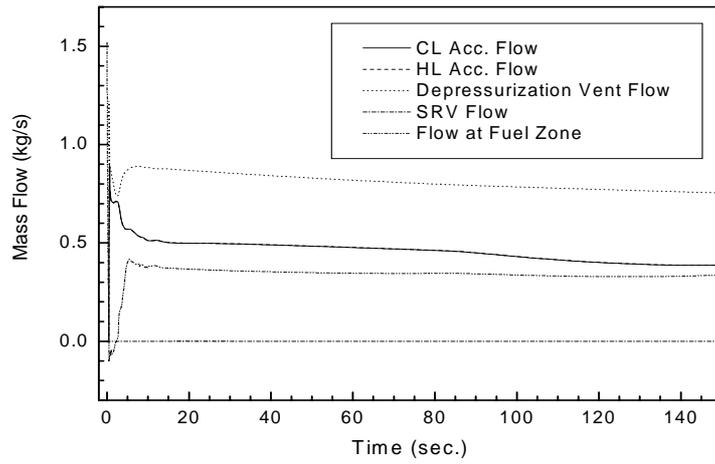
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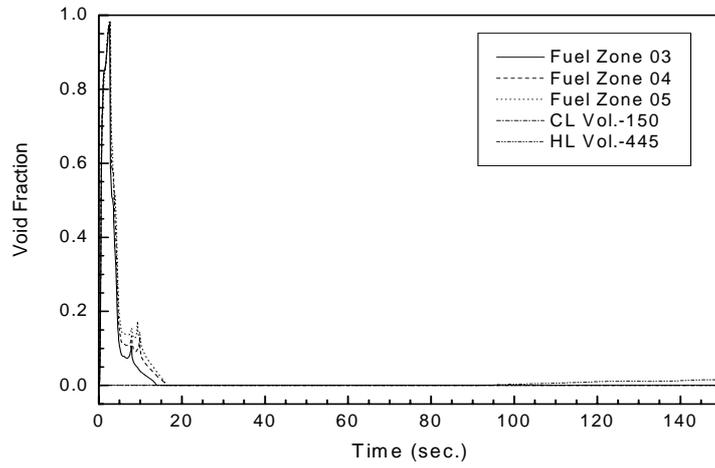
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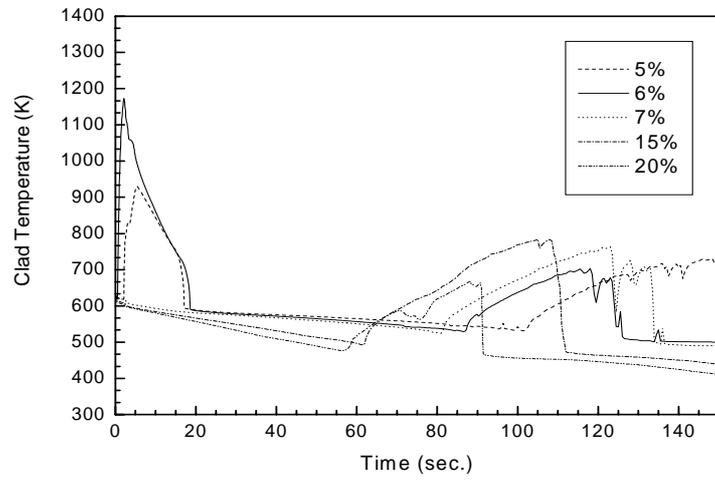
6. 1



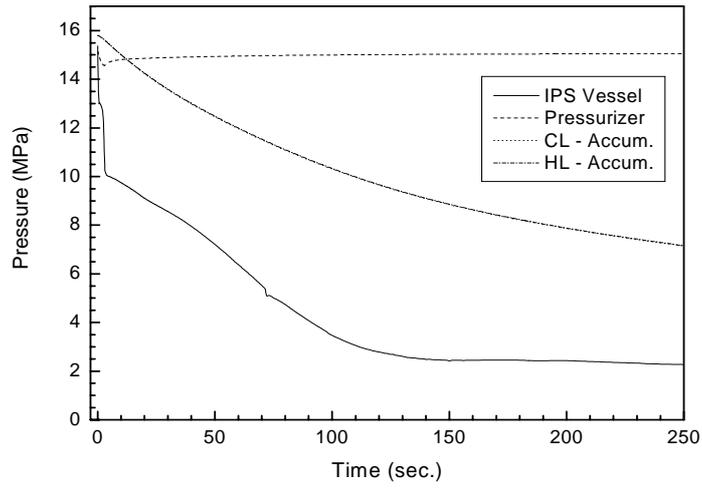
7. 1



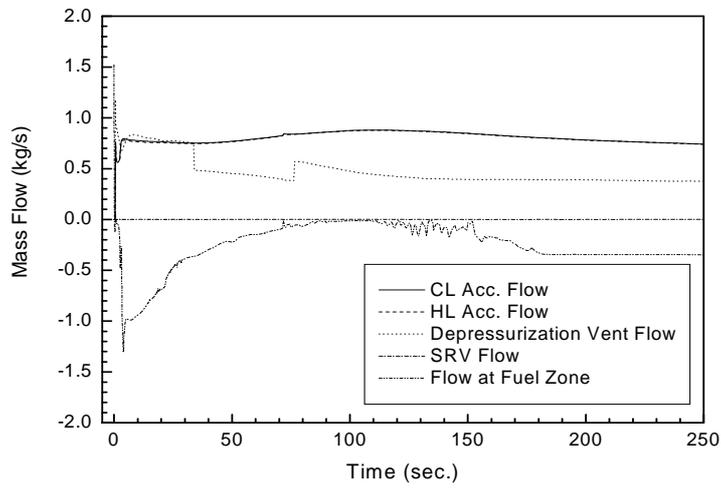
8. 1



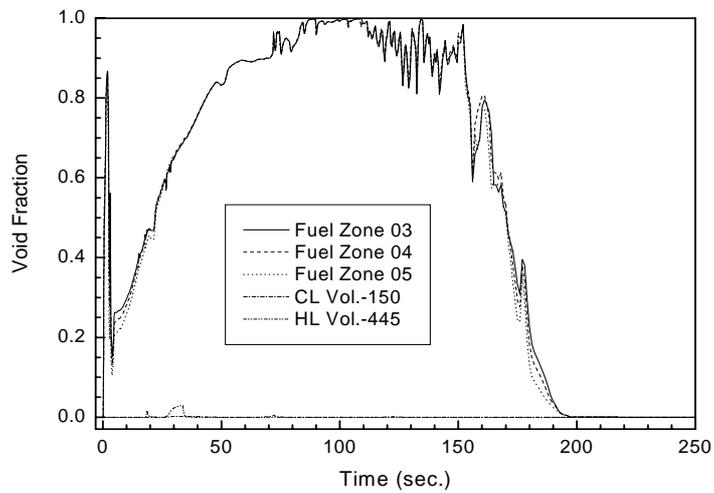
9.



10.



11.



12.