# Preliminary Analysis of Supercritical CO2 loop Using MARS

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## 1. Introduction

The supercritical carbon dioxide Brayton cycle has become a strong candidate for use in the next generation nuclear reactors. The simplicity, compactness, cost, and thermal efficiency are known as the main advantages of the cycle. The relatively high efficiency can be reached in lower temperature range than that of helium Brayton cycles. The size of the turbomachinery is also very small compared to the helium Brayton cycle or the steam Rankine cycle, which also contributes in reducing the cost. [1]

To test the cycle under many different conditions a facility named SCO2PE has been built in KAIST. The experimental data is to be compared with simulation results from a code named MARS.

## 2. Tools

### 2.1 SCO<sub>2</sub>PE [2]

 $SCO_2PE$  is a facility built to test  $CO_2$  under supercritical conditions. This device is designed specifically to observe how a compressor performs near the critical point of  $CO_2$ . Although there is a lot of data regarding pumps for water or compressors for ideal gas, there is still insufficient data for use with supercritical  $CO_2$ . The device consists of a heat exchanger, a canned motor pump type compressor, and a globe valve.



Fig. 1. A schematic diagram of the main CO<sub>2</sub> loop

## 2.2 MARS Code [3]

MARS Code is a thermal hydraulic system analysis code developed by KAERI. It was designed as an integration of RELAP5/MOD3 and COBRA-TF codes. It is mainly designed for the analysis of water but CO<sub>2</sub>, D<sub>2</sub>O, He, Be, and Na can also be tested.

## 3. Methods and Results

The compressor was modeled with MARS first individually. The experimental results were compared to those from MARS. After the validation of the compressor, the whole loop was analyzed.

#### 3.1 Compressor

The pump component in MARS is used for modeling the S-CO<sub>2</sub> compressor. It requires the homologous curve for accurate analysis. Four important variables are required for the homologous curve: volumetric flow, head rise, rotational speed, and shaft torque. These values are then compared to the rated values and the ratio is added to the input. The following formulas were used to calculate the missing values.

H=w/g (1)  $w=\Delta enthalpy$  (2)  $\tau=Power/\omega$  (3)

Table I: Rated Values

| $H_R$     | 44.85 m                   |
|-----------|---------------------------|
| $	au_R$   | 28.11 <i>N</i> · <i>m</i> |
| ωr        | 483.8 <i>m³/s</i>         |
| $ ho_{R}$ | $321.9 \ kg/m^3$          |
| $Q_R$     | $4.893 E - 3 m^3/s$       |

| $\nu = Q/Q_R(4)$                |  |
|---------------------------------|--|
| $\alpha = \omega / \omega_R(5)$ |  |
| $h=H/H_R(6)$                    |  |
| $\beta = \tau / \tau_R(7)$      |  |

Table 2: Homologous Curve Values

| $\nu/lpha$ | $h/\alpha^2$ | $\beta/\alpha^2$ |
|------------|--------------|------------------|
| 0.0        | 0.0          | 0.0              |
| 0.657      | 3.179        | 0.906            |
| 1.0        | 1.0          | 1.0              |

Table 3: Homologous Curve Values

| $\alpha/\nu$ | $h/v^2$ | $\beta/\nu^2$ |
|--------------|---------|---------------|
| 0.0          | 0.0     | 0.0           |
| 0.657        | 0.0508  | 0.604         |
| 1.0          | 1.0     | 1.0           |

Two tests with different boundary conditions were executed, one with a pressure boundary on the inlet and outlet, the other with a mass flow rate boundary on the inlet and a pressure boundary on the outlet.

### Table 4: Pressure Boundary Result

|            | Mass Flow Rate |
|------------|----------------|
| Experiment | 1.3417 kg/s    |
| MARS       | 1.1020 kg/s    |

Inlet Pressure: 8.020 MPa Outlet Pressure: 8.271 MPa Inlet Temperature: 39.43°C Outlet Temperature: 42.68°C Mass Flow Rate Difference: 17.87%

## Table 5: Mass Flow Rate Boundary Result

|                | Experiment | MARS      |
|----------------|------------|-----------|
| Inlet P        | 8.020 MPa  | 8.160 MPa |
| Pressure Ratio | 1.031      | 1.014     |
| Inlet T        | 39.43°C    | 41.04°C   |
| Outlet T       | 42.68°C    | 43.69°C   |

Outlet Pressure: 8.271 MPa Mass Flow Rate: 1.3420 kg/s Pressure Ratio Difference: 1.65%  $\Delta$  P Difference: 55.78%

## 3.2 Loop

The SCO<sub>2</sub>PE is nodalized as follows.



Fig. 2. Nodalization Model of SCO<sub>2</sub>PE

Total length of the pipe without the compressor and heat exchanger is 9.983 m and pipe area is 1.140e-3 m<sup>2</sup>. A time dependent volume was added to stabilize the pressure of the system. Form loss coefficients were given as the following table.

| Table 6: Form L | loss Coefficient |
|-----------------|------------------|
|-----------------|------------------|

| Threaded 90° elbow | 1.5   |
|--------------------|-------|
| Globe Valve        | 101.0 |
| Flow Meter         | 7.0   |

The heat exchanger was nodalized with a pipe for the tube containing  $CO_2$  and a time dependent volume for the shell containing water. The pipe was divided into 10 volumes and properties of stainless steel 316 was used. The specification of the heat exchanger is as follows.

Table 7: Heat Exchanger Specification

| Area         | 9.428e-4 m <sup>2</sup> |
|--------------|-------------------------|
| Total Length | 21.221 m                |
| Form Loss    | 34.2                    |

Results of the loop computation is as following.

|                |   | Exp.      | MARS      | Diff.  |
|----------------|---|-----------|-----------|--------|
| Pump           | Р | 8.315 MPa | 8.309 MPa | 0.07%  |
| Inlet          | Т | 312.99 K  | 313.80 K  | 0.26%  |
| Pump           | Р | 8.608 MPa | 8.611 MPa | 0.03%  |
| Outlet         | Т | 315.12 K  | 317.40 K  | 0.72%  |
| HX Inlet       | Р | 8.406 MPa | 8.408 MPa | 0.02%  |
|                | Т | 314.01 K  | 315.46 K  | 0.46%  |
| HX Outlet      | Р | 8.323 MPa | 8.315 MPa | 0.10%  |
|                | Т | 312.99 K  | 313.90 K  | 0.29%  |
| Mass Flow Rate |   | 1.5725    | 1.1617    | 26.15% |
|                |   | kg/s      | kg/s      |        |
| Power          |   | 10.70 kW  | 9.63 kW   | 10.0%  |

### 4. Summary and Further Works

A MARS base input deck was created for SCO<sub>2</sub>PE. The compressor performance was checked by comparing experimental results with computational results from MARS. Then the loop performance was validated preliminarily.

Since MARS is mainly constructed for two phase flow modeling and the correlation package is based on water, we should check how well current MARS can simulate supercritical  $CO_2$  system. Any alternatives that can draw more accurate results should be checked for. Also, additional experiments should be executed in different temperature and pressure ranges.

## REFERENCES

 [1] Dostal, V., 2004. A supercritical carbon dioxide cycle for next generation nuclear reactors, Dissertation, Massachusetts Institute of Technology, Department of Nuclear Engineering.
 [2] Jekyoung Lee, Jeong Ik Lee, Yoonhan Ahn, Seong Gu Kim, Jae Eun Cha, SCO2PE OPERATING EXPERIENCE ANS VALIDATION AND VERIFICATION OF KAIST\_TMDA, ASME Turbo Expo, Jun.3-7, 2013, Texas, USA [3] Bub-Dong Chung, Kyung-Doo Kim, Sung-won Bae, Jae-Jun Jeong, Seung Wook Lee, Moon-Kyu Hwang, Churl Yoon, MARS CODE MANUAL VOLUME I, KAERI, Feb. 2010.