# Thermal Design for 2nd Irradiation Test of SFR Nuclear Fuel

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## 1. Introduction

A capsule was designed for the second irradiation test of SFR nuclear fuel. Design for the first irradiation test of SFR nuclear fuel was conducted in 2009 [1], and the second test of double the burn-up is under planning. In performing a worthy irradiation test, keeping the temperature of the capsule stable during the irradiation period is essential. To predict the temperature distribution in the capsule, a temperature analysis is performed using the thermal analysis code. In this project, we attempted a thermal analysis for a stability verification of the capsule. The objectives of this study are as follows.

A. Determining the linear power for the cladding target temperature.

B. Confirmation of the nuclear fuel melting temperature and ONB (Onset of Nucleate Boiling) temperature at the sealing tube surface.

C. Accident analysis on the locked rotor event and control rod withdrawal accident.

The capsule was designed based on the results obtained through the analysis.

## 2. Capsule model

#### 2.1 Modeling

The capsule will be irradiated in the OR5 test hole of HANARO at 30 MW up to a burn-up 6 at%. Fig. 1 shows the cross section of the fuel rod surrounded by a sealing tube. In this study, it was considered that only the fuel rod channel of the coolant is passed. A gap between the cladding and sealing tube was fixed at 0.06 mm and 0.07 mm. the gap filled with helium gas. The gap between the cladding and the fuel was filled with sodium. The cladding is T92, and the sealing tube is Stainless Steel 316L. The fuel is determined by U-Zr and U-Zr-5Ce. The specifications of the fuel rod are listed in Table 1. Depending on its diameter, fuel is divided into two types. Fig. 2 shows the full body of the SFR irradiation capsule. Six fuel rods are respectively loaded at the top and bottom.





Fig. 2. The full body of SFR irradiation capsule.

Ten	Rac	dius	Dian	Thickness	
<u>10p</u>	in	out	in	out	THICKNESS
Fuel		2.81		5.62	
Inner	3.00	3.50	6.00	7.00	0.50
gap	3.50	3.56	7.00	7.12	0.06~0.07
Outer	3.56	4.06	7.12	8.12	0.50
					(mm)
Pottom	Rac	dius	Diam	Thicknoss	
Bottom	in	out	in	out	THICKHESS
Fuel		1.99		3.98	
Inner	2.25	2.75	4.50	5.50	0.50
gap	2.75	2.81	5.50	5.62	0.06~0.07
Outer	2.81	3.31	5.62	6.62	0.50

Table 1: The dimensions of the fuel rod in the axial direction.

\* The gap can be changed to 0.07 mm.

\* Outer diameter of the fuel rod channel: 11.5 mm

## 2.2 Analysis

The temperature distribution of the model of an irradiation test capsule is estimated using the 1-dimensional heat transfer code, GENGTC [2]. The GENGTC code is calculated by considering the structure in the radius direction and the heat generation. The temperature distribution and linear power of the fuel were calculated based on the cladding temperature. Considerations of this design are as follows [3].

A. The target temperature of the cladding is 650  $\,^\circ\!\!\mathbb{C}$  under a steady state.

B. Fuel center T: < melting point (1235  $^{\circ}$ C)

C. Sealing tube surface T: < ONB temp. (125  $^{\circ}$ C)

To simulate an accident, the temperatures were calculated after raising the linear power to 131% (the control rod withdrawal accident) and lowering the flow rate to 56% (the locked rotor event) [4]. Fig. 3 shows the out pile test system. The velocities of the fuel rod channel can be found through the out pile hydraulic test results, and the convective heat transfer coefficients are calculated using these values. The flow cross section of channel is different according to the vertical position, so the flow velocity of the fuel rod channel at the top is 10.96 m/s and at the bottom is 8.2 m/s.



Fig 3. The out-pile test system.

The heat transfer coefficient at the outer surface of the sealing tube is 2.08 to 4.45 W/cm<sup>2</sup>.  $^{\circ}$ C. Under a locked rotor event, it is 2.06 to 3.28 W/cm<sup>2</sup>.  $^{\circ}$ C. The temperature of the cooling water in the reactor in-core is about 40  $^{\circ}$ C. Details related to the modifications of the cladding are not considered.

#### 3. Results

Tables 2 and 3 shows the temperature analysis results of the fuel rod channel. The center temperatures of the nuclear fuel were confirmed to not exceed the melting point including under an accident situation. In a locked rotor event (flow rate 56%), the surface temperatures of the sealing tube exceeded the ONB temperature at the bottom position. The linear power in a 0.06 mm gap is 400 to 634 W/m, in the transient analysis is 525 to 831 W/m, in a 0.07 mm gap is 371 to 546 W/m, and in the transient analysis is 456 to 716 W/m. We could see that the linear power in the 0.06 mm gap case is higher than in the 0.07 mm gap case. The thermal distribution satisfies the temperature stability in the 0.07 mm gap case. However, the temperature distribution of the first irradiation test capsule is lower than the calculated value, and a high linear power reduces the period of irradiation.

Fable	2:	The	tem	perat	ure	dist	ibu	tion	and	line	ar	power	ſ
calcula	ated	using	the	GEN	GTC	C cod	le (ł	neliur	n gaj	o.0 c	6 m	ım).	
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A. Standard (Target temperature 650°C)					
	Te	Linear			
(Position)	Fuel	Inner	Tube	power	
( ,	center	cladding	surface	(W/cm)	
U-Zr	048.0	640.0	027	506	
(Top)	940.9	049.9	52.7	550	
U-Zr		GEO E	102.9	400	
(Bottom)	956.5	050.5	102.8	400	
U-Zr-5Ce	052.4	640.7	027	624	
(Тор)	955.4	049.7	92.7	054	
U-Zr-5Ce	960 5	640 5	102.7	426	
(Bottom)	900.5	049.5	102.7	420	
Convection heat transfer coefficient					

: 4.45 (3.28) W/cm<sup>2</sup>.°C

B. Flow rate 56%

Fuel Type	Te	Linear		
(Position)	Fuel	Inner	Tube	power
(. osidon)	center	cladding	surface	(W/cm)
U-Zr	072.1	674.2	1227	FOG
(Top)	975.1	074.5	125.7	590
U-Zr	085.7	678.1	140.0	400
(Bottom)	905.7	078.1	140.0	400
U-Zr-5Ce	077.4	672.0	1227	624
(Top)	977.4	075.0	125.7	054
U-Zr-5Ce	088.2	677.4	120.8	426
(Bottom)	900.Z	077.4	139.0	420
Convection	heat transfe	er coefficient		

: 2.80 (2.06) W/cm<sup>2</sup>.°C

C. Power 131%

	Te	Linear			
(Position)	Fuel center	Inner cladding	Tube surface	power (W/cm)	
U-Zr (Top)	1151.2	762.4	108.9	781	
U-Zr (Bottom)	1170.7	770.1	122.2	525	
U-Zr-5Ce (Top)	1157.9	762.2	108.9	831	
U-Zr-5Ce (Bottom)	1174.3	769.1	122.0	558	

Convection heat transfer coefficient

: 4.45 (3.28) W/cm<sup>2</sup>.°C

Table	3:	The	temperature	distribution	and	nnear	power
calcula	ated	using	the GENGTO	C code (heliu	m gap	0.07 n	ım).

A. Standard (Target temperature 650 °C)						
Fuel Type	Te	Linear				
(Position)	Fuel center	Inner cladding	Tube	power (W/cm)		
U-Zr (Top)	908.7	650.3	85.3	514		
U-Zr (Bottom)	918.2	650.1	94.6	348		
U-Zr-5Ce (Top)	911.9	649.8	85.2	546		
U-Zr-5Ce (Bottom)	921.3	650.0	94.5	371		

Convection heat transfer coefficient

: 4.46 (3.28) W/cm<sup>2</sup>.°C

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B. Flow rate 56%

Fuel Type	Te	Linear			
(Position)	Fuel	Inner cladding	Tube	power (W/cm)	
U-Zr (Top)	928.8	670.7	112.1	514	
U-Zr (Bottom)	941.3	673.2	126.4	348	
U-Zr-5Ce (Top)	931.8	669.8	112.0	546	
U-Zr-5Ce (Bottom)	943.8	672.7	126.3	371	

Convection heat transfer coefficient

: 2.80 (2.07) W/cm<sup>2</sup>.°C

C. Power 131%

Fuel Type	Te	°C)	Linear		
(Position)	Fuel center	Inner cladding	Tube surface	power (W/cm)	
U-Zr (Top)	1099.4	763.4	99.2	673	
U-Zr (Bottom)	1119.5	770.1	111.3	456	
U-Zr-5Ce (Top)	1104.5	762.9	99.1	716	
U-Zr-5Ce (Bottom)	1123.5	769.7	111.3	486	

Convection heat transfer coefficient

: 4.46 (3.28) W/cm<sup>2</sup>.°C

## 4. Conclusions

To conduct a verification of the SFR nuclear fuel, irradiation testing is required. For this test, a stable capsule design is essential. A thermal design of the capsule was carried out for the second irradiation test of SFR nuclear fuel. Owing to the high temperature of the bottom position, caution is needed when designing the capsule. In consideration of the linear power and temperature of the cladding, it was decided to design the gap to be 0.06 mm. After determining the linear power of the nuclear fuel with reference to the results, the thickness of the Hf shielding material put in the capsule will be determined. The Hf shielding material serves as a controller of the neutron flux coming into the capsule.

## REFERENCES

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