# Application of 3D Printing Process for Fabrication of ZrO2-Zr2Cu Composite Fuel

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# 1. Introduction

Nuclear fuel is one of the most important components in nuclear applications considering safety and economy. Oxide fuel like UO<sub>2</sub> is used for commercial nuclear reactors among a variety of nuclear fuel types (metal, silicide, nitride, etc.). UO2 fuel has a low thermal conductivity which results in detrimental effects in terms of plant safety, performance, etc. Thermal conductivity of the nuclear fuel could be improved by development of the composite fuel which contains a metal matrix. In previous studies, a variety of the composite fuels were studied such as UO2-W and UO2-Fe [1-2]. UO<sub>2</sub> particles in the composite fuel fabricated by hot isostatic pressing, spark plasma sintering, etc. could not be uniformly distributed. A uniform distribution of the fuel particles is mightily important as a composite fuel because it results in hot spots and ultimately leads to failure of the nuclear fuel.

In this study, additive manufacturing (AM) with subsequent pressureless melt infiltration was introduced to fabricate the composite fuel with a uniform distribution and various forms. Preform was fabricated using  $ZrO_2$  powder as a surrogate of  $UO_2$  by binder jet process. The melt infiltration using  $Zr_2Cu$  alloy was carried out as a follow-up process to fabricate the composite fuel. Additionally, microstructure of the  $ZrO_2$ - $Zr_2Cu$  composite fuel was investigated to evaluate the fabrication of the composite fuel.

#### 2. Methods and Results

As shown in Fig. 1,  $ZrO_2$  preforms with a cubic (10 x 10 x 10 mm) and cylinder (D10 x h10 mm) forms were printed by binder jet processing. Several processing parameters for printing the preform were listed in Table 1.  $ZrO_2$  powders with a diameter of 30 µm (D50) were used and it was stacked with a thickness of 100 µm. The preforms were shaped as cubic and cylinder forms by saturating solvent-based binder and curried in an oven. Key parameters of the preform fabrication process by binder jet processing were identified as powder density, layer thickness, and binder saturation.



Fig. 1. ZrO<sub>2</sub> preforms printed by binder jet processing; (a) cubic and (b) cylinder forms

Table 1. Processing parameters of binder jet processed ZrO<sub>2</sub> preforms

Processing parameters	
Material	ZrO <sub>2</sub>
Particle size	30 μm (D50)
Forms	Cubic, Cylinder
Binder	Aquafuse (Exone)
Desired saturation	40%
Layer thickness	100 µm

As shown in Fig. 2, the preform was heated up to 1200 °C for debinding and melt infiltration, and it was cooled in the furnace.  $Zr_2Cu$  alloy as a matrix material was used to infiltrate into the porous preform.  $Zr_2Cu$  alloy was spontaneously infiltrated from the bottom of the preform, and the composite was fabricated up to approximately 50% of the preform height. Only  $ZrO_2$  particles with keeping the shape of preform remained in the uninfiltrated region.



Fig. 2. Debinding/infiltration profile for binder jet processed ZrO<sub>2</sub> preform

Microstructure of the  $ZrO_2-Zr_2Cu$  composite was investigated by scanning electron microscopy (SEM). As shown in Fig. 3,  $Zr_2Cu$  alloy was infiltrated into the porous preform. A phenomenon in which Zr segregated to the surface of  $ZrO_2$  particles was observed while the  $Zr_2Cu$  alloy was infiltrated into the  $ZrO_2$  preform, and it was speculated that there would be an interaction at the interface between the  $ZrO_2$  and  $Zr_2Cu$  during the infiltration process.



Fig. 3. SEM-EDS images of ZrO2-Zr2cu composite

## 3. Conclusions

In this study,  $ZrO_2$ - $Zr_2Cu$  composite fuel was fabricated by applying a 3D printing process with the subsequent melt infiltration process.  $Zr_2Cu$  alloy could be spontaneously infiltrated into the  $ZrO_2$  porous preform up to approximately 50% of the preform height. As a result, the fabrication of the composite fuel was positively evaluated. However, it is needed to improve infiltration by controlling the surface tension of matrix alloy or interfacial energy between matrix alloy and preform.

## REFERENCES

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